



Helicopter Fatigue Life Monitoring in the 21st Century

Symposium on
Condition-Based Maintenance
for Highly Engineered Systems

A.A. ten Have

National Aerospace Laboratory NLR

Structures and Materials Division

tel: + 31 527 248292

fax: +31 527 248210

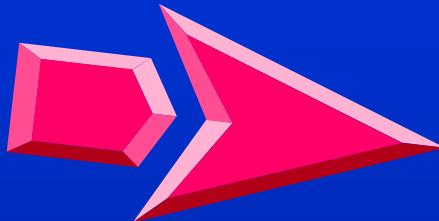
e-mail: have@nlr.nl

Pisa, It.

September 25-26-27, 2000

Life Cycle

OPERATIONAL
USAGE



DEGRADATION

time
loads
environment
(corrosion, erosion,
humidity,
temp. etc.)

REMEDIES

sched. maintenance
periodical inspections
operational limitations
replacements
repairs/mods
finite lives



...Safety,
Economy,
Readiness...

Fatigue Life

- **On-Condition Items**

- * 'Damage Tolerant' design
- * periodical inspections and repairs (if necessary)
- * non-finite fatigue life
- * finite economical life
- * inspection intervals based on operational usage estimate
(done by manufacturer)

- **Lifed Items**

- * finite 'Safe Life'
- * no in-service inspections
- * based on statistics (99%/95%-rule)
- * finite life based on operational usage estimate
(done by manufacturer)

Fatigue Life



Typical for
helicopter
components

- **Lifed Items**

- * finite 'Safe Life'
- * no in-service inspections
- * based on statistics (99%/95%-rule)
- * finite life based on operational usage estimate
(done by manufacturer)

Fatigue Life calculation

material testing

- full scale components
 - sub components
 - coupon specimens

reduction factor for variance in testresults

'safe life' in number of cycles

design loads

- mission type/mix
 - events per mission
 - loads per event

reduction factor for variance in usage

safe 'exchange rate' in cycles/FH

**Safe
Life
(FH)**

Fatigue Life calculation

material testing

- full scale components
 - sub components
 - coupon specimens

reduction factor for variance in testresults

'safe life' in number of cycles

**Safe
Life
(FH)**



reduction factor for variance in usage

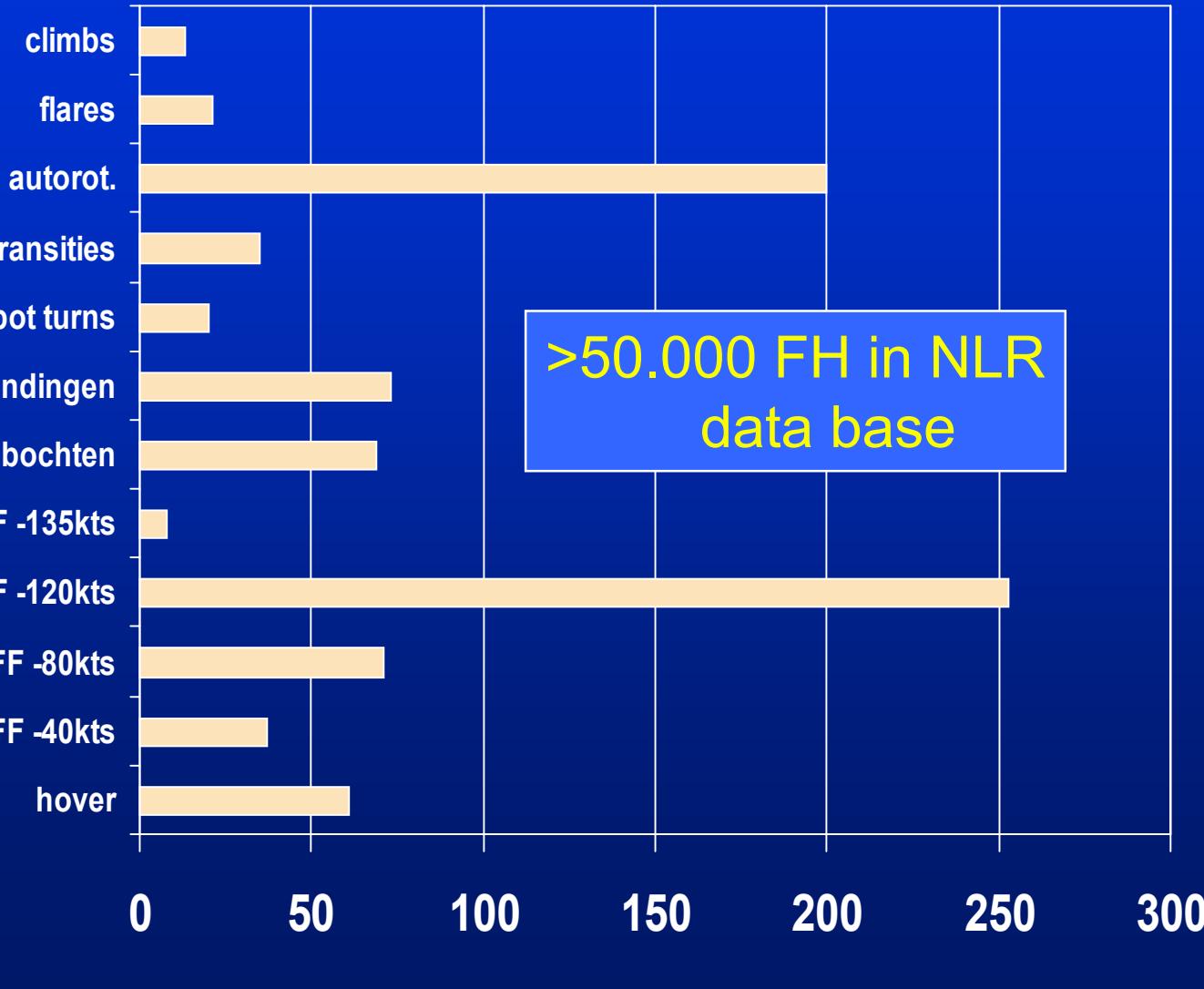
safe 'exchange rate' in cycles/FH



RNLN Fatigue Life Monitoring...

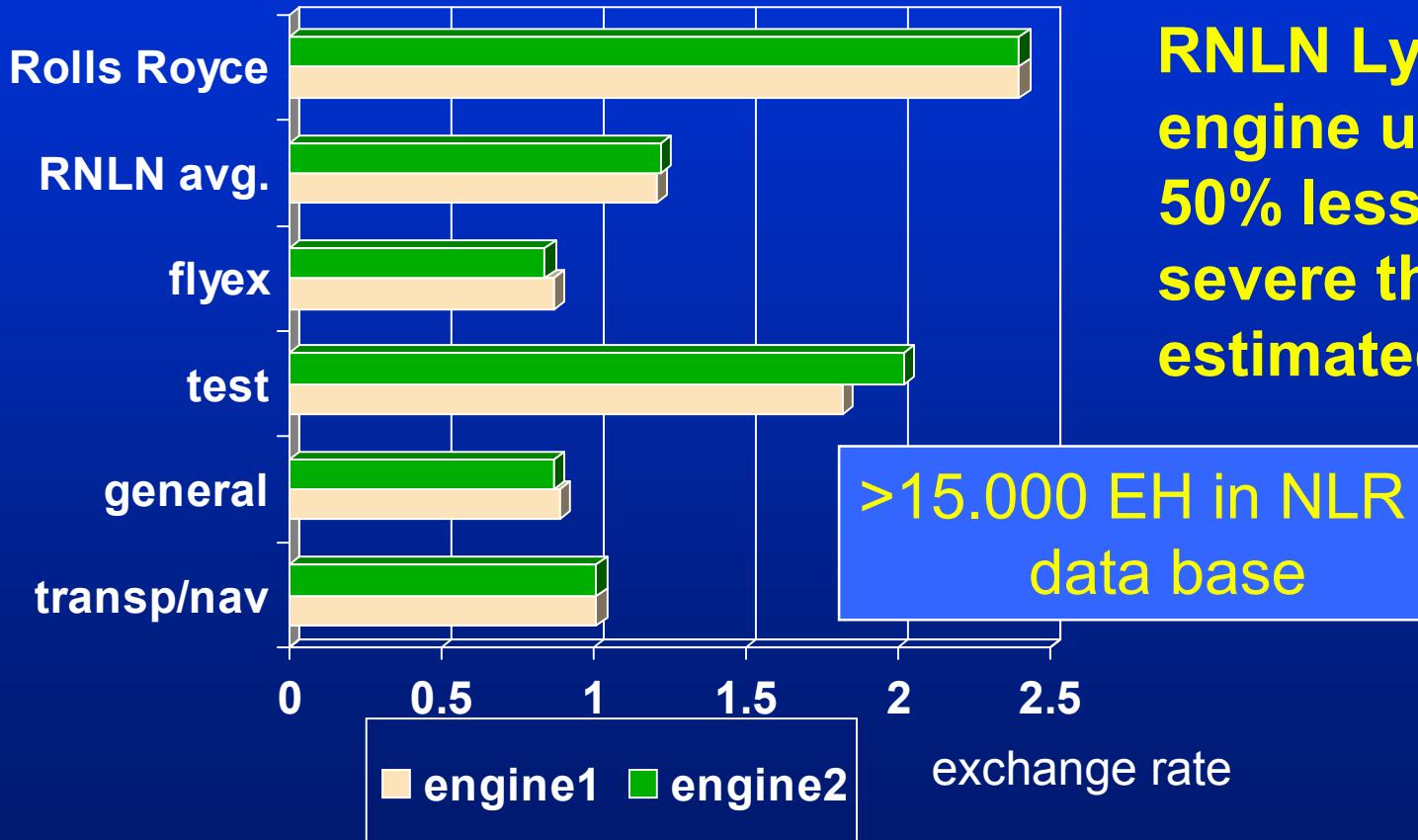
- Lynx Usage Monitoring
- Lynx Gem Cyclic Life Control
 - * Rolls Royce Gem engine
- Lynx AIDA
 - * multichannel retrofit system

RNLN Lynx Usage Monitoring



- Less high-speed flight
- less hover time
- less # Indngs
- lower ratio deck/land
- more autorot.
- more avg speed FF

RNLN Gem Cyclic Life Control



RNLN Lynx AIDA

● Nr monitoring

- * audio/visual warnings
- * signal: 1 x RPM

● Gem engine Cyclic Life Control

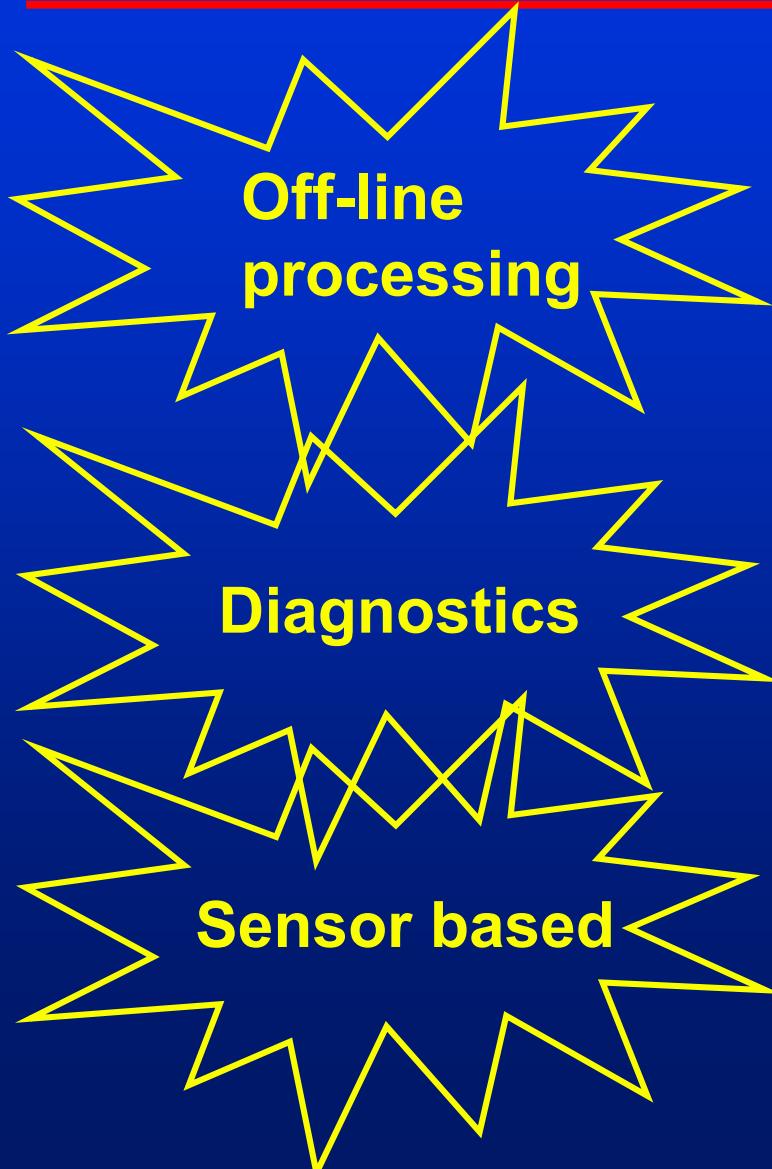
- * Individual Aircraft Tracking
- * Rolls-Royce Service Bulletin
- * signals: 4 x RPM ($N_{h,1,2} + N_{f,1,2}$)

● Usage Monitoring

- * CLUMS data-base
- * lifting frame, sponson
- * signals:
 - 2 discretes (WoW, RadAlt)
 - 15 analogue (alt, speed, φ , Nr, Nh, Nf, Tq, Sponson strains, spares)



Fatigue Life Monitoring - Now -

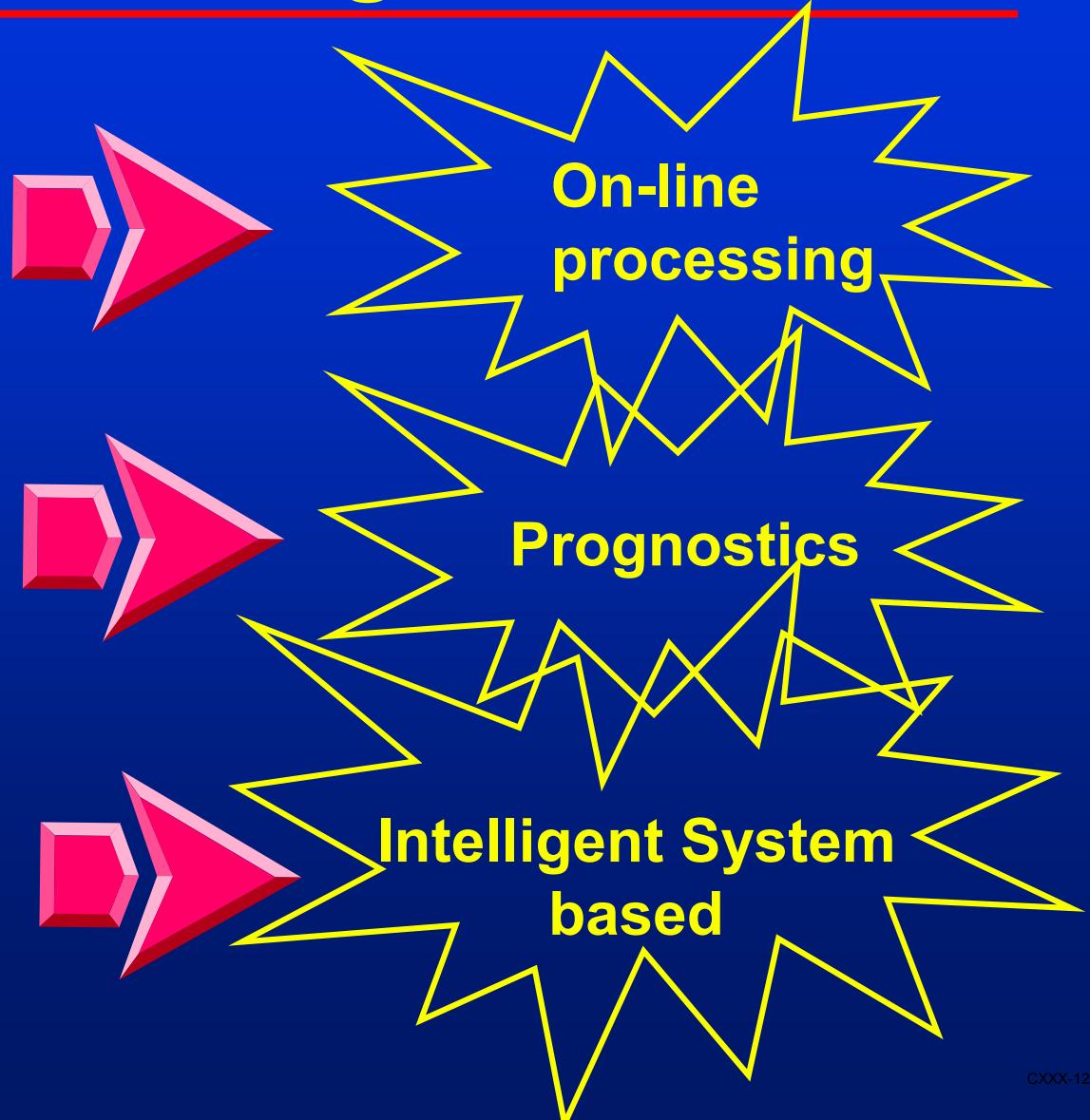


Fatigue Life Monitoring -Transition-

**Off-line
processing**

Diagnostics

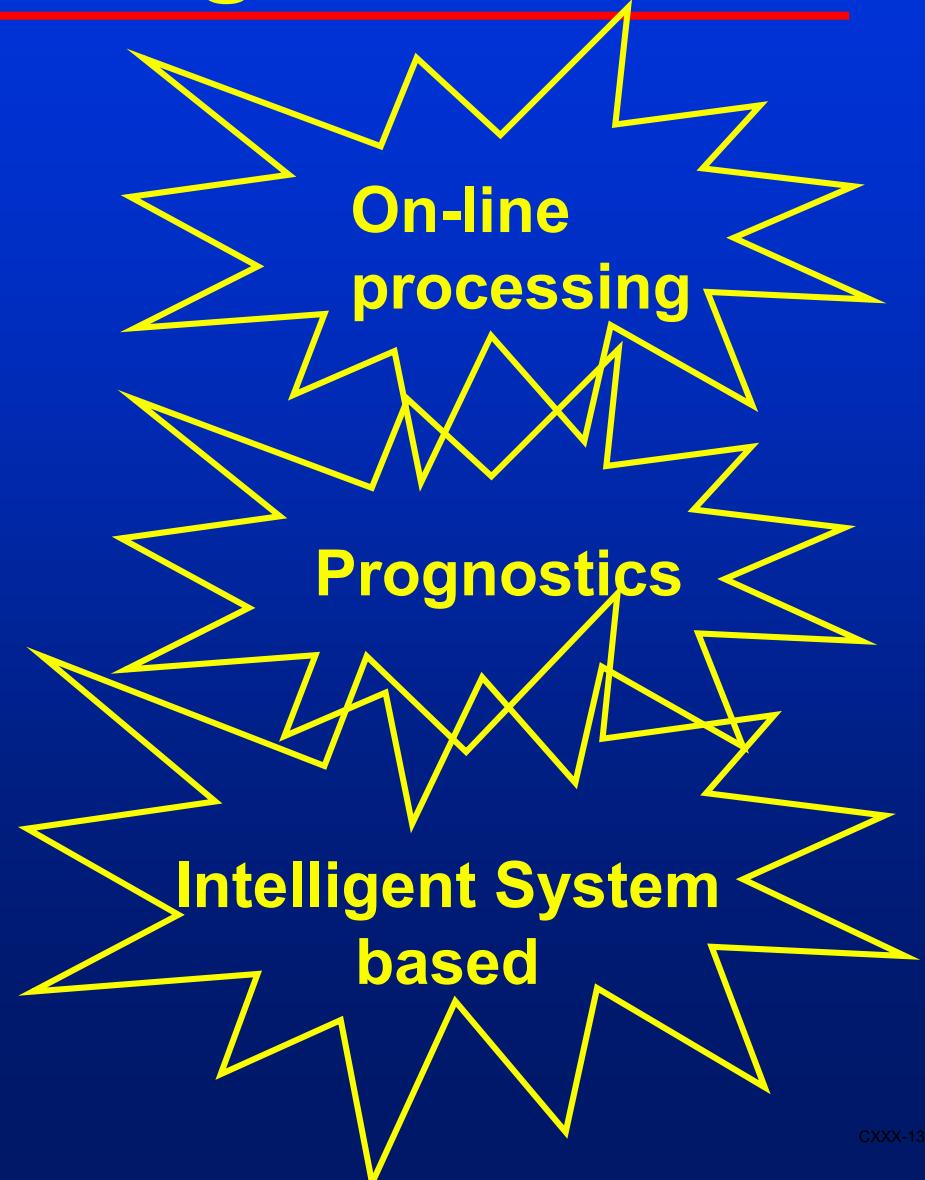
Sensor based



Fatigue Life Monitoring - Future -

Prognostics and Health Management (PHM) systems

- Intelligent Load Monitoring ILM
 - virtual strain sensors
 - neural networks
- Data Mining
 - autom. regime recognition
 - engine prognostics
- Reasoners
 - model, rule, case based
 - genetic algorithms



ILM example - Now -



Sponson
bending

Sponson
torsion

Off-line Assessment of:

- **Landing Type**
 - deck/land
 - harpoon Y/N
- **Landing Regime**
 - running
 - MPOG
 - T.O. and Landing Weight statistics
- **Limit Load Exceed.**
- **Vibration Level**
- **Damage accumulation**

ILM example - Future -



PHM will enable:

- Elimination of strain sensors**
- Strains derived from configuration, flight and environmental data**
- On-line calculation of accumulated damage**
- Prognostics for fleet and mission planning**

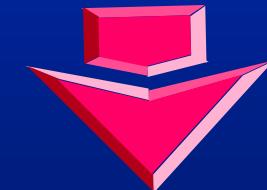
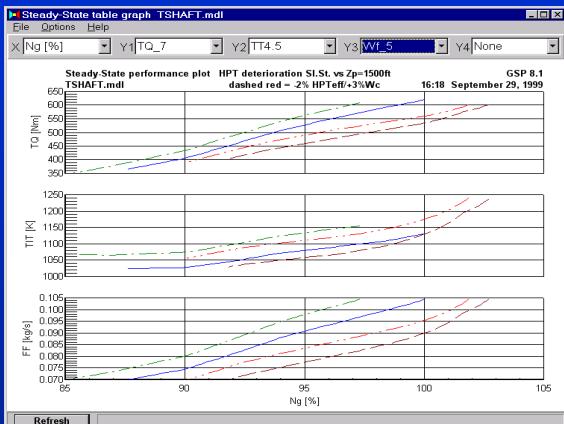
Engine prognostics

Problem: HPT Deterioration
Sea-Level efficiency - 2% with
massflow +3%

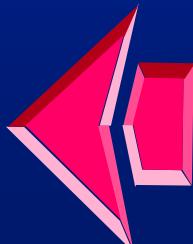
**Deterioration effect is hidden
due to flight condition
changes (causing variation
in performance data)**



GSP Turboshaft model

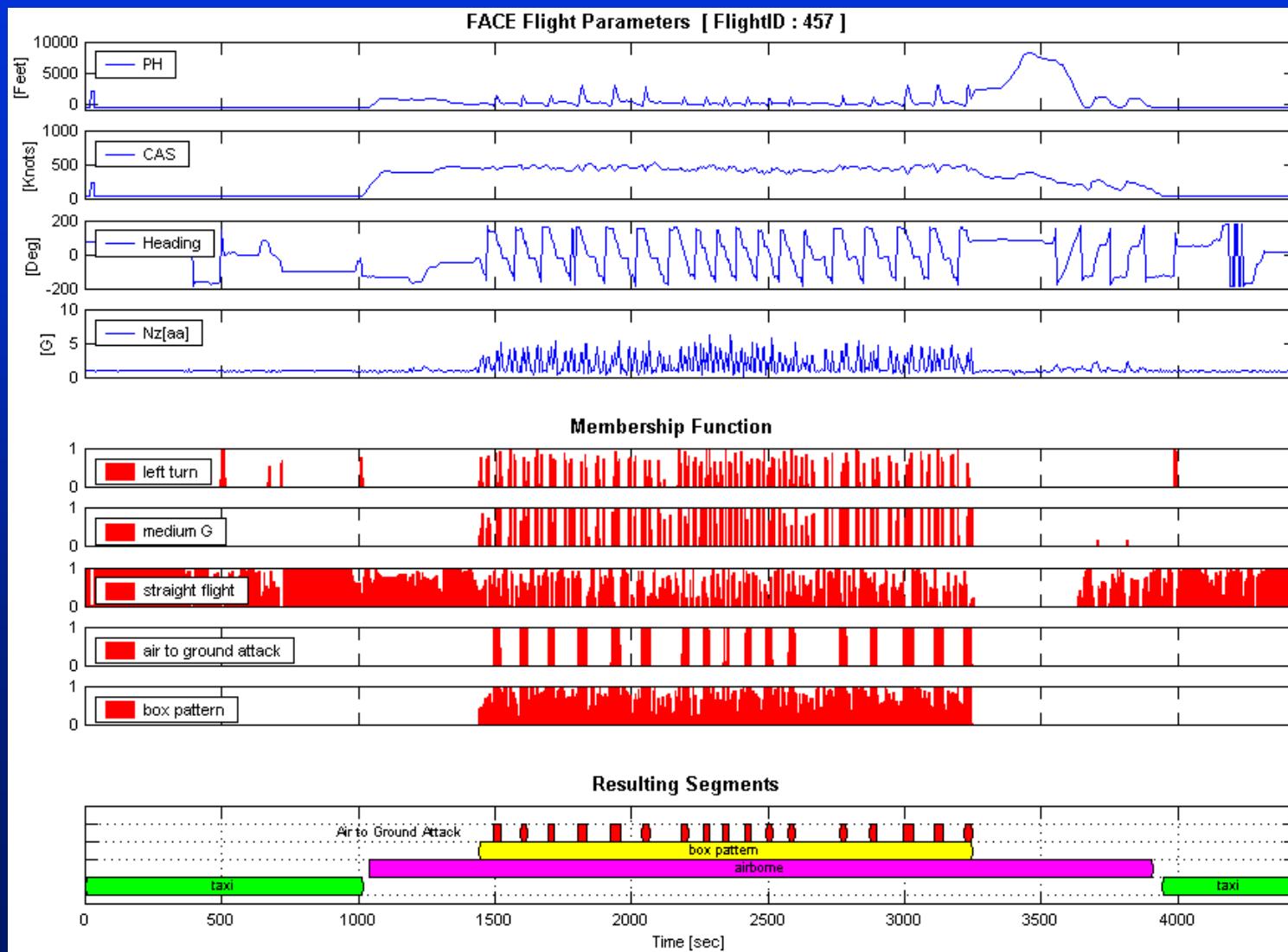


**Prediction of
Deterioration Effects**



**Data Mining finds Known
AND Unknown relations**

Data Mining example - Fighter -



Finally

- Better understand 20th century technology as a basis for future developments
- 21st century goal: Transition of emerging fixed-wing PHM technologies to the rotary-wing environment
- R&D Structural Integrity \$\$ will shift from the traditional

Mechanical and Materials Engineering

to

IT, Quality Assurance and Certification areas