

Alitalia

Cbm Oriented Engines Fleet Maintenance Plan

Pisa, September 26th, 2000

The current Alitalia Engines Fleet (slides 2-3) is composed by qty 334 Operating Turbofan Engines and it is maturing an average of 519K engine yearly cycles corresponding to a more than 250K Fleet Revenue Flights per year; this mission is supported by a total of 60 spare engines.

The engines are in the thrust rating from 20000 to 60690 Lbs and are of General Electric, Pratt & Whithney and CFMI (G.E. – SNECMA Consortium) manufacture; the oldest entered the fleet in 1978 (CF6-50E2/B747-200), the youngest in 1999 (CFM56-5B4P/A320).

All the engines in the fleet (slide 4) are being managed under Maintenance Programs oriented to CBM Philosophy; these programs are customized by engine/aircraft model and continuously adjusted in looking for the best reliability of the propulsion system and in the respect of the Air Carrier maintenance cost saving criteria. The Engineering Department of Alitalia takes the responsibility of such a continuos activity.

The Engine Maintenance Program includes time based and upon-condition detail tasks to perform; these are generally inspections, checks, tests and are mostly performed by NDT practices; some external engine components are also soft time based replaced for bench test and then repaired or overhauled if-when necessary. The Life Limited Rotating Parts and Airworthiness Directives are those imposing Hard Time Limits requiring scheduled removals of engines; on spot special hard time maintenance tasks could also be part of the Maintenance Plan like in the case of OEM recommended Campaign Change or Special Inspection Program; in these cases, the application of the maintenance special task will be upon the modification standard of the individual engine in the fleet.

As part of the Engine on-condition Maintenance Program (slide 5), Alitalia Engineering implemented and is still using ECM and SAGE diagnostic tools; these are on wing engine monitoring devices developed by PWA and G.E./SNECMA, respectively, and offered to customers as additional engine equipment. The main advantage from operating both the ECM and SAGE in preventing engine malfunctions and premature failures could be directly measured in the reduced aircraft delays and cancellations as well as in the reduced engine maintenance material cost. Performances of these computer diagnostic tools are of similar capability; the most recent G.E. Release of SAGE is here presented in more details.

The SAGE program (slides 6-7-8) is performed by means of data retrieval from magneto-optical supports (OQAR) and, as an additional method, by crew instrumental manual readings (IRD); however, the IRD is for cruise data recording only.

At each aircraft transit to main-bases, OQAR supports are removed and sent to AZ Computer Center at FCO; the main frame computer recognizes the logical flight criteria needed to extract take-off and stable cruise data; the cruise and takeoff data are then downloaded directly into PC/Workstation for SAGE computing; the SAGE outputs are analyzed by engineer on daily basis.

The SAGE measures the main engine parameters deviations versus expected trend and gas generator lines; these deviations are computed at takeoff and cruise phases of each flight during which also the vibration levels of both low and high pressure rotor groups are recorded and in trend plotting presented; at takeoff, the SAGE computes the individual engine Exhaust Gas Temperature

(EGT) and provides the relevant Margin to Takeoff E.G.T. Red Line (e.g. 960 °C for CF6 Engines); the Outside Air Temperature Limit (OATL) at which the engine is able to perform a full power takeoff by not exceeding the EGT Red Line is also computed by the SAGE.

The Cruise Trend Report (CTR) shows (slide 9) how much the EGT, F/F and N2 (high press spool expressed as percentage of RPM) deviate from the gas generator line and from the initialized reference value; it also reports the fan vibration level “V” and the core vibration level “R”, as well as the lubricating oil Pressure PRS (psi) and Temperature TMP (°C). The data are presented in both Raw and Smoothed Mode for better evaluating the actual single flight data and the relevant deviation from the previous trend; all the data are bias N1.

The Takeoff Trend Report (TTR) shows (slide 10-11) both the Raw and Smoothed single takeoff EGT Margin and relevant OATL; the actual N1K and the percentage of Thrust Derating (FNDR) and N1 Derating (N1DR) are also reported per each takeoff.

Both CTR and TTR are provided with an Alerting Device; it counts how many parameter exceeded the reference value (alerting windows) during cruise or takeoff; details on such occurrences can be analyzed into SAGE data base; the alerting windows can be customized by the operator.

The tight relationship (slide 12) between the Takeoff EGT Margin (EGTM), OATL and Engine Time Since Installation (TSI) allows the engineer to predict the potential residual life of the individual engine in the fleet; the residual on wing engine life is computed by limiting the OATL at the corner point value (e.g. 30 °C for CF6-50E2,-80C2) corresponding to a “Zero EGT Engine”.

When applied on the entire engine fleet (slide 13-14), the Time to Zero EGT computation allows to build up the figure on the potential short-long term removals forecast allowing a better plan of the shop activity and avoiding unacceptable engine removals concentration.

An additional way to monitor the engine vibration levels (slide 15-16) is provided by SAGE with the Last Fifty Takeoff Data Report. This report, the Fan Vib and the Core Vib are being presented for analysis on both raw and smoothed trend plotting.

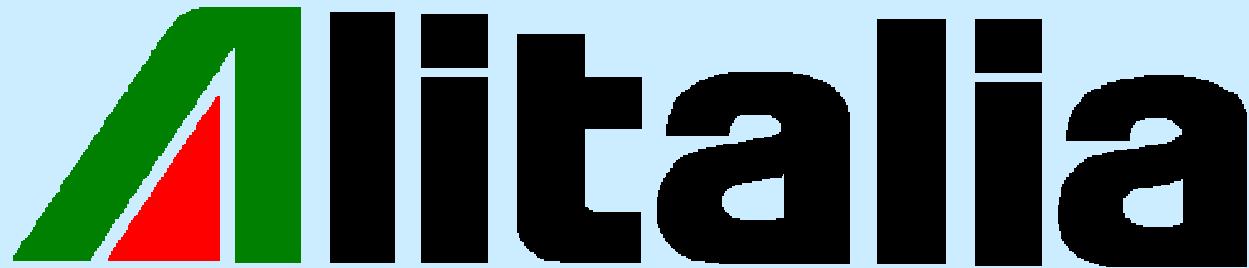
The Vib Data of both Cruise and Takeoff Reports are being stored in a dedicated Data Base for any further technical analysis; a long term Vib Trend can be of great help to the engineer when it is necessary to verify the effectiveness of a Vib trouble shooting corrective action applied on the engine.

If compared to the past (slide 17-18), the workscope management during an Engine Shop Visit (ESV) today is based more on engineering parameters; the single main engine group (module) workscope can be identified and the best combination of these groups on marshalling can be achieved. The improved modularity design, that the most modern engines provide, allows the implementation of this concept; the availability of more reliable data detailing the status of the engine under shop visit provide a very useful decision tool to define the workscope level to be applied.

FCO, Sep1/00

G. de Filippis

A. Tateo



ENGINES FLEET

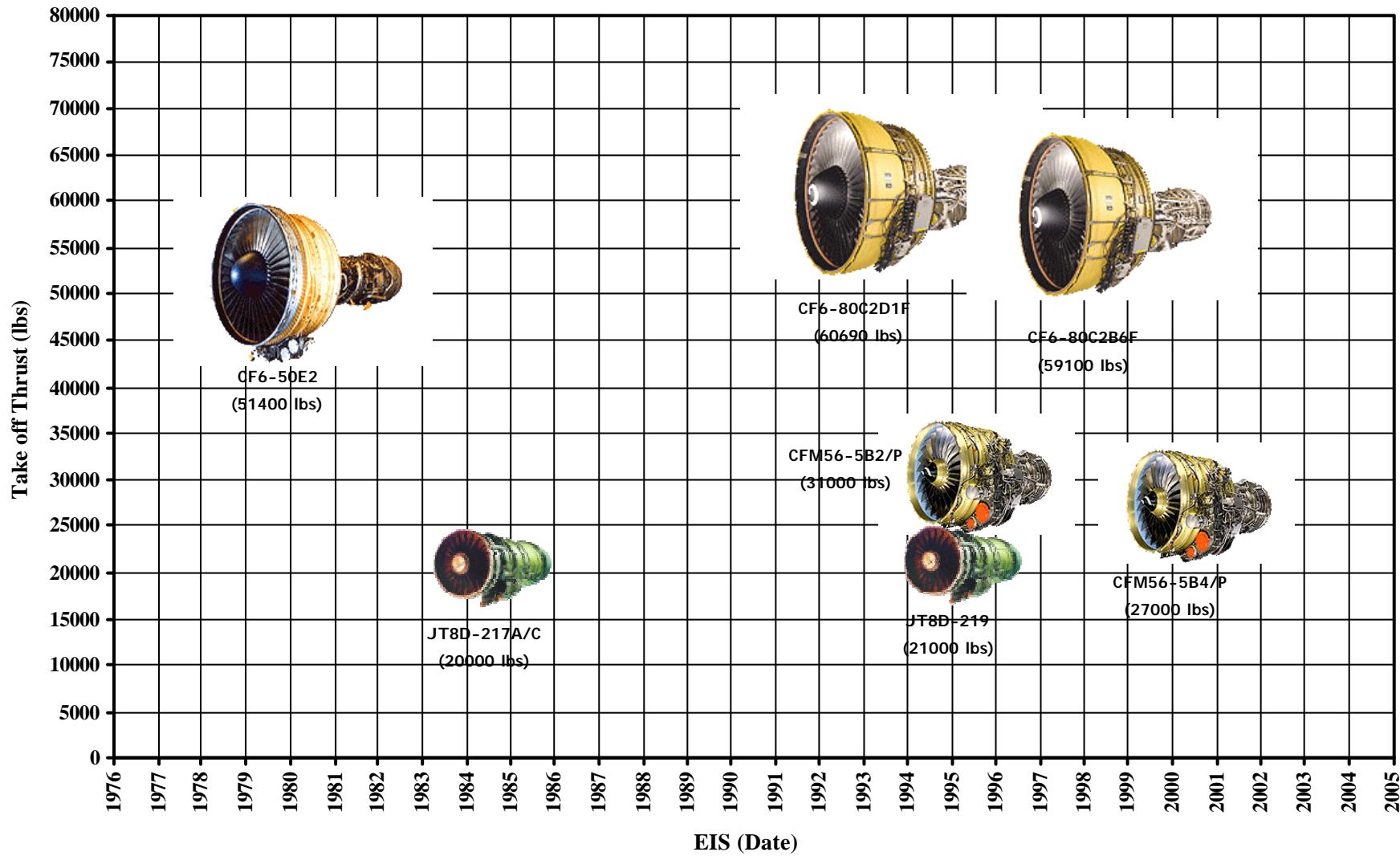
C.B.M. ORIENTED MAINTENANCE PLAN

Pisa, September 26th, 2000



Alitalia - Current Engine Families*

* all but regional engines

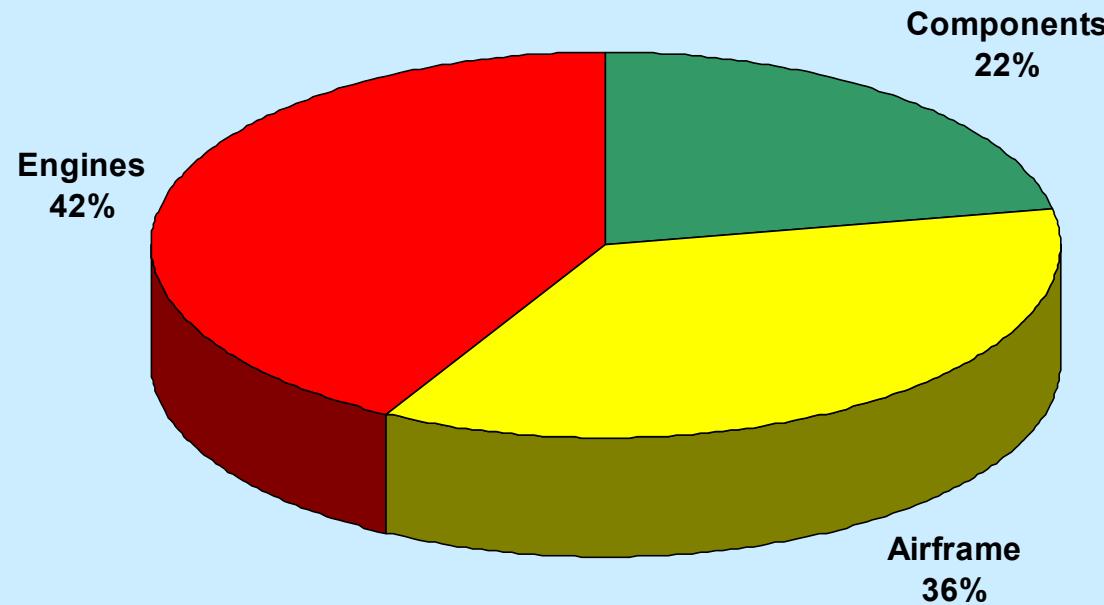



- Current Engines Fleet *

<u>model</u>	<u>thrust (lbs)</u>	<u>a/c</u>	<u>EIS</u>	<u>Cys x Year = eng fleet</u>	<u>a/c fleet</u>	<u>f/l</u>
JT8D-217A/C	20000	MD82	dec83	2026x2x 89= 360628	180314	1.0
JT8D-219	21000	MD83	mar94	1079x2x 5= 10790	5395	2.3
CFM56-5B2/P	31000	A321	mar94	1618x2x23= 74428	37214	1.6
CFM56-5B4/P	27000	A320	mar99	1900x2x 6= 22800	11400	2.0
CF6-50E2	51400	B747	oct78	546x4x10= 21840	5460	7.1
CF6-80C2D1F	60690	MD11	dec91	540x3x 8= 12960	4320	9.0
CF6-80C2B6F	59100	B767	nov96	661x2x12= 15864	7932	7.2

* all but regional fleet

Aircraft Maintenance Cost - Breakdown (typical)



The Engine Maintenance Cost is the most impacting AMC



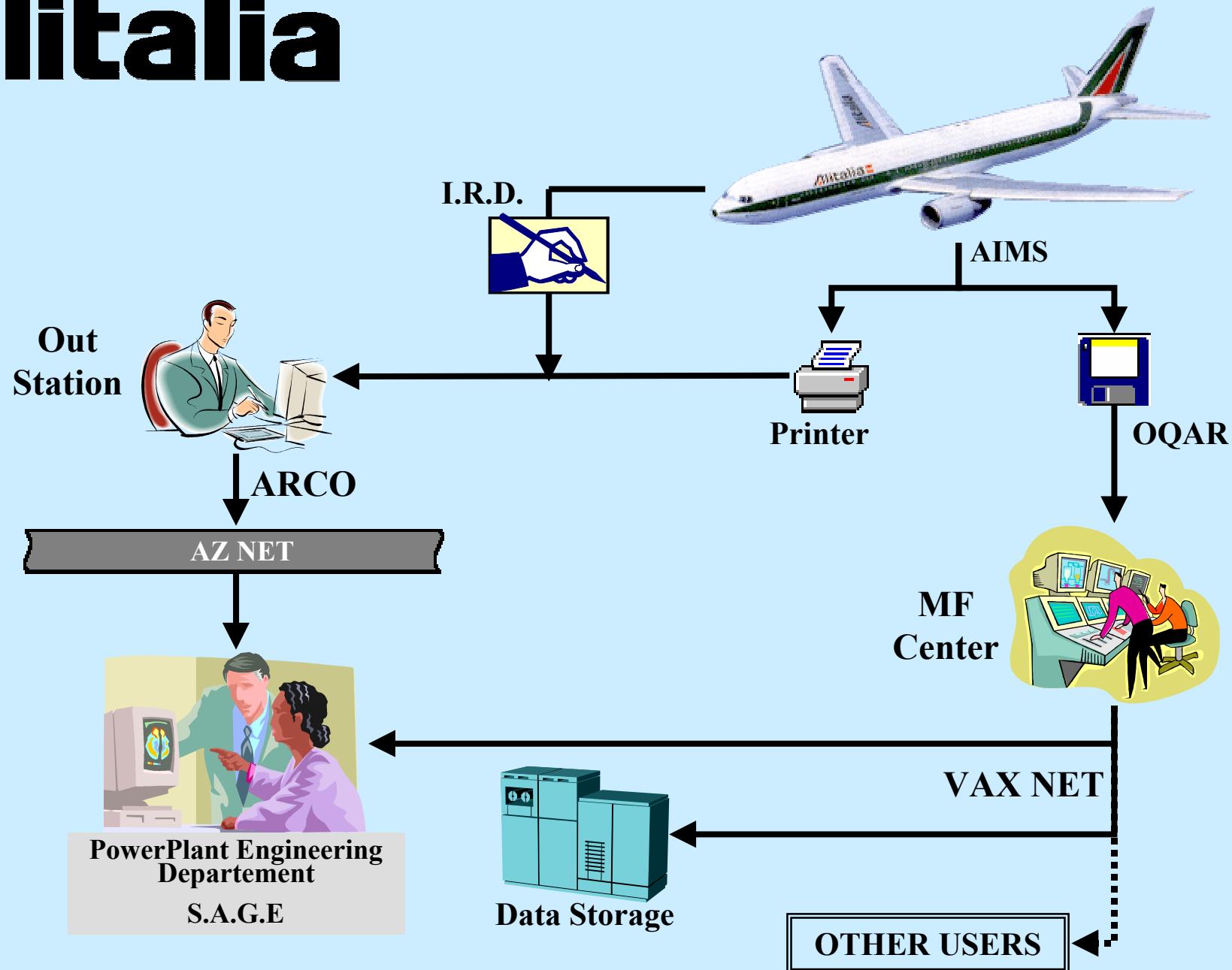
System for Analysis of Gas turbine Engines (SAGE)

The SAGE is the most helpful tool in supporting the “on-condition based maintenance plan” for General Electric & CFMI engines.

The SAGE software is OEM developed and it is offered to operators as an additional engine equipment.

The SAGE provides a method of:

- monitoring the individual on wing eng performance deterioration
- predicting the relevant remaining time to shopvisit
- checking the individual eng rotor unbalance level
- preventing eng in service failure
- saving engine maintenance cost





STABLE CRUISE

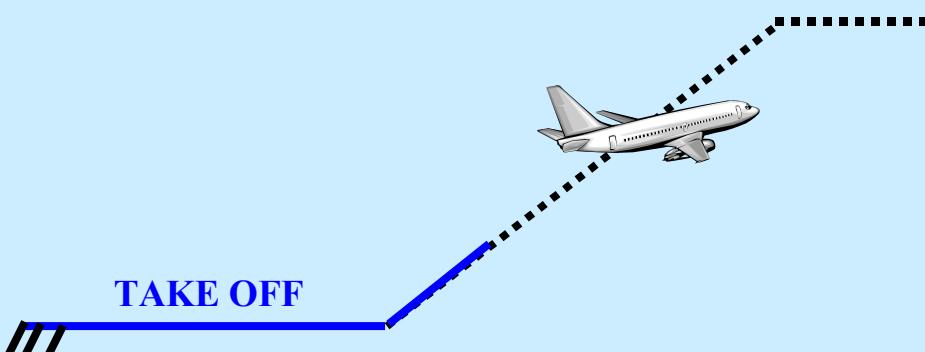
- A. Verify following conditions for 2 minutes minimum...:

PARAMETER	Tolerance Range
ALT > 25000 Feet	+/- 100 Feet
81.5 < (N2#1 and N2#2) < 98.1	+/- 1%
TAT	+/- 1°C
0.6 < MACH < 0.9	+/- 0.005
N1 and N2 of engines	+/- 1%
ENGAICE1 and ENGAICE2	= OFF (0)
WAICEL and WAICER	= OFF (0)
ISOV1 and ISOV2	= CLOSED (0)

- B. ... then the following conditions for 60 seconds minimum:

PARAMETER	Tolerance Range
ALT > 25000 Feet	+/- 100 Feet
81.5 < (N2#1 and N2#2) < 98.1	+/- 0.5%
TAT	+/- 0.5°C
0.6 > MACH < 0.9	+/- 0.004
N1 and N2 of engines	+/- 0.5%
ENGAICE1 and ENGAICE2	= OFF (0)
WAICEL and WAICER	= OFF (0)
ISOV1 and ISOV2	= CLOSED (0)

STABLE CRUISE //



TAKE OFF

TAKE OFF DATA

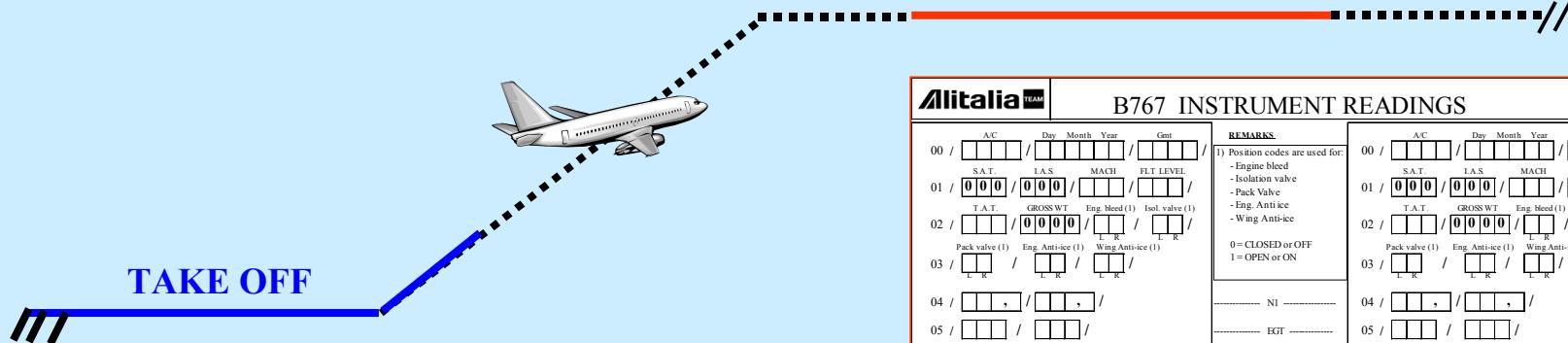
During the period of 60 seconds following the CAS has reached 100 Kts, identify the highest EGT peak value between engines; all data but logical should be recorded during the period of 2 seconds before-after the EGT Peak ref point; the average 4 seconds data are computed for OAT'L.

STABLE CRUISE REPORT

00/ ACI D' DATA/ GMT/
 01/ SAT/ CAS/ MACH/ ALT/
 02/ TAT/ GROSSWT/ ENGBLD1/ ENGBLD2/ I SOV1/ I SOV2/
 03/ PACK1/ PACK2/ ENGAI CE1/ ENGAI CE2/ WAI CEL/ WAI CER/
 04/ NI#1/ NI#2/
 05/ EGT#1/ EGT#2/
 06/ N2#1/ N2#2/
 07/ FF#1/ FF#2/
 08/ PLA#1/ PLA#2/
 09/ OI LP#1/ OI LP#2/
 10/ OI LT#1/ OI LT#2/
 11/ VI BF#1/ VI BF#2/
 12/ VI BT#1/ VI BT#2/

STABLE CRUISE REPORT

00/ B767/ 120497/ 1642/
 01/ -40/ 283/ 796/ 330/
 02/ -11/ 1469/ 11/ 00/
 03/ 11/ 00/ 00/
 04/ 0919/ 0919/
 05/ 656/ 654/
 06/ 0979/ 0978/
 07/ 2545/ 2502/
 08/ 00/ 00/
 09/ 56/ 54/
 10/ 108/ 112/
 11/ 07/ 02/
 12/ 01/ 02/

STABLE CRUISE

TAKE OFF REPORT

00/ ACI D' DATA/ GMT/
 01/ SAT/ CAS/ MACH/ ALT/
 02/ TAT/ GROSSWT/ ENGBLD1/ ENGBLD2/ I SOV1/ I SOV2/
 03/ PACK1/ PACK2/ ENGAI CE1/ ENGAI CE2/ WAI CEL/ WAI CER/
 04/ NI#1/ NI#2/
 05/ EGT#1/ EGT#2/
 06/ N2#1/ N2#2/
 07/ FF#1/ FF#2/
 08/ PLA#1/ PLA#2/
 09/ OI LP#1/ OI LP#2/
 10/ OI LT#1/ OI LT#2/
 11/ VI BF#1/ VI BF#2/
 12/ VI BT#1/ VI BT#2/
 13/ 00/ 00/
 14/ 00/ 00/

TAKE OFF REPORT

00/ B767/ 120497/ 1342/
 01/ 006/ 190/ 284/ 001/
 02/ 010/ 1346/ 11/ 00/
 03/ 11/ 00/ 00/
 04/ 1051/ 1051/
 05/ 920/ 931/
 06/ 1032/ 1031/
 07/ 8991/ 9002/
 08/ 00/ 00/
 09/ 63/ 64/
 10/ 075/ 072/
 11/ 02/ 03/
 12/ 08/ 04/
 13/ 00/ 00/
 14/ 00/ 00/

B767 INSTRUMENT READINGS				PAGE No.
A/C	Day	Month	Year	Gmt
00 / <input type="text"/> / <input type="text"/> / <input type="text"/> / <input type="text"/> /	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>
S.A.T.	L.A.S.	MACH	FLT LEVEL	
01 / <input type="text"/> / <input type="text"/> / <input type="text"/> / <input type="text"/> /	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>
T.A.T.	GROSS WT.	Eng bleed (1)	Isol. valve (1)	
02 / <input type="text"/> / <input type="text"/> / <input type="text"/> / <input type="text"/> /	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>
Pack valve (1)	Eng. Anti-ice (1)	Wing Anti-ice (1)		
03 / <input type="text"/> / <input type="text"/> / <input type="text"/> / <input type="text"/> /	<input type="text"/>	<input type="text"/>	<input type="text"/>	<input type="text"/>
<input type="text"/> R	<input type="text"/> R	<input type="text"/> R	<input type="text"/> R	
04 / <input type="text"/> , / <input type="text"/> , /				
05 / <input type="text"/> / <input type="text"/> /				
06 / <input type="text"/> , / <input type="text"/> , /				
07 / <input type="text"/> 0 / <input type="text"/> 0 /				
08 / <input type="text"/> 0 / <input type="text"/> 0 /				
09 / <input type="text"/> / <input type="text"/> /				
10 / <input type="text"/> / <input type="text"/> /				
11 / <input type="text"/> , / <input type="text"/> , /				
12 / <input type="text"/> , / <input type="text"/> , /				
REMARKS:				
1) Position codes are used for: -Engine bleed -Isolation valve -Pack Valve -Eng. Anti ice -Wing Anti-ice				
0=CLOSED or OFF 1=OPEN or ON				
N1 -----				
EGT -----				
N2 -----				
F/F -----				
Throttle position (rear size)				
OIL PRESS. -----				
OIL TEMP. -----				
FAN VIBS. -----				
TURB VIBS. -----				
Notes:				
Flight	Number	From	To	
Capt.				Signatures C/M-2

REPORT ID: CRTRND

GE ENGINE CONDITION MONITORING PROGRAM
CRUISE PERFORMANCE MONITORING - MOST RECENT RECORDS

SAGE V3.0.3 PC - Jan 99

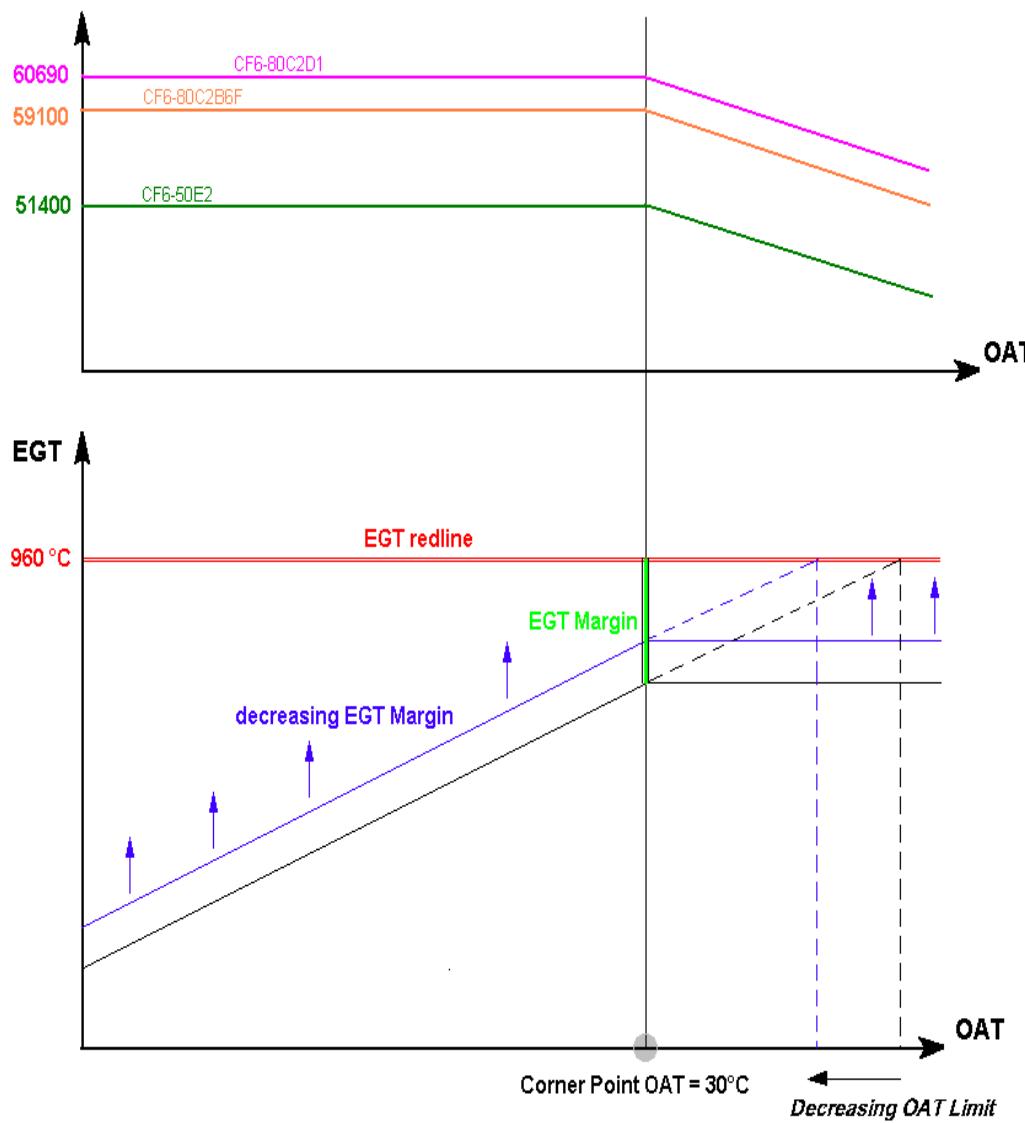
REPORT DATE: 8/22/ 0
PAGE: 1

	AIRTPY	ENGTYP	ENGSN	INSDAT	N1MOD	TCC	FNRAT	CONFIG	CONTROL	SELECTOR	(SCALES VARIED)
AAAA	-1	B767-300ER	CF6-80C2B6F	XXX182	990309	5	*	*****			
SMOOTH		-30..-20..-10...EGT..10..20..30				-1....X....0....X....N2....X....2....X....3	CRZ	OIL OIL MAINT ALT			
DATE	VIB..1....2....3			0....2....4....F/F....8....10....12		3....4....VSV....6....7		SLOATL TMP PRS CODES CTR			
31499A	R=	0.8	V=	0.3	-8.8		4.61	VSV=-999.00	N2=	1.49	
31499B	R=	0.8	V=	0.3	-8.8		4.61	VSV=-999.00	N2=	1.49	
40100C	. V R	.	G	.		F	.	X	2		
50100C	. V R	.	G	.		F	.	X	2		
60100C	. V R	.	G	.		F	.	X	2		
70100C	. V R	.	G	.		F	.	X	2		
80900	. V R	.	G	.		F	.	X	2	118 57.	0
80900	.	*	.	G	.	F	.	X	2	112 56.	0
80900	.	*	.	G	.	F	.	X	2	102 57.	0
81000	.	*	.	G	.	F	.	X	2	108 56.	0
81000	.	*	.	G	.	F	.	X	2	112 55.	0
81100	.	VR	.	G	.	F	.	X	2	104 56.	0
81100	.	*	.	G	.	F	.	X	2	112 56.	0
81100	.	*	.	G	.	F	.	X	2	111 57.	0
81100	.	*	.	G	.	F	.	X	2	89 55.	0
81200	.	*	.	G	.	F	.	X	2	104 56.	0
81200	.	VR	.	G	.	F	.	X	2	100 55.	0
81300	.	*	.	G	.	F	.	X	2	70 57.	0
81300	.	*	.	G	.	F	.	X	2	80 55.	0
81300	.	*	.	G	.	F	.	X	2	112 54.	0
81300	.	VR	.	G	.	F	.	X	2	112 54.	0
81300	.	V R	.	G	.	F	.	X	2	108 53.	0
81400	.	V R	.	G	.	F	.	X	2	119 55.	0
81400	.	VR	.	G	.	F	.	X	2	116 54.	0
81400	.	VR	.	G	.	F	.	X	2	103 57.	0
81400	.	VR	.	G	.	F	.	X	2	108 55.	0
81500	.	VR	.	G	.	F	.	X	2	104 56.	0
81500	.	VR	.	G	.	F	.	X	2	112 55.	0
81500	.	VR	.	G	.	F	.	X	2	102 56.	0
81500	.	VR	.	G	.	F	.	X	2	108 56.	0
81500	.	VR	.	G	.	F	.	X	2	104 56.	0
81500	.	VR	.	G	.	F	.	X	2	114 56.	0
81800	.	*	.	G	.	F	.	X	2	118 56.	0
81900	.	*	.	G	.	F	.	X	2	108 58.	0
81900	.	*	.	G	.	F	.	X	2	110 57.	0
81900	.	*	.	G	.	F	.	X	2	102 55.	0
82000	.	*	.	G	.	F	.	X	2	111 58.	0
82000	.	*	.	G	.	F	.	X	2	104 56.	0
82000=	.	*	.	G	.	F	.	X	2	103 58.	0
82000=	.	*	.	G	.	F	.	X	2	97 56.	0
RAW DATA		-30..-20..-10...EGT..10..20..30				-1....X....0....X....N2....X....2....X....3	CRZ	MODULE CHG ALT			
DATE	VIB..1....2....3			0....2....4....F/F....8....10....12		3....4....VSV....6....7		SLOATL MAINT CODES CTR			
81500	*	.	G	.		F	.	X	2		0
81600	V R	.	G	.		F	.	X	2		0
81800	* V	X	X				X	X			0
81900	* V	X	X				X	X			0
81900	*	G	.			F	.	X	2		0
82000	* V	.	G	.		F	.	X	2		0
82000	* V	X	X				X	X			0
82000=	V R	.	G	.		F	.	X	2		0
82000=	V R	.	G	.		F	.	X	2		0

1

REPORT ID: TKTRND		GE ENGINE CONDITION MONITORING PROGRAM TAKEOFF PERFORMANCE MONITORING - MOST RECENT RECORDS						SAGE V3.0.3 PC - Jan 99			REPORT DATE:	8/22/ 0		
											PAGE:	1		
AAAA	-1	AIRTYP	ENGYP	ENGSN	INSDAT	N1MOD	TCC	FNRAT	CONFIG	CONTROL	SELECTOR	(SCALES VARIED)		
SMOOTH		B767-300ER	CF6-80C2B6F	XXX182	990309	5	*	*****		N2	ACTUAL	MAINT ALT		
DATE	CSI			0...10...20...MAR..40...50...60						0....5...MAR..15...20	N1K	FNDR	N1DR	CODES CTR
31499A				46.10		43.74				- 999.00				
31499B				46.10		43.74				- 999.00				
30800C	.	H	.		O		X							
40100C	.	H	.		O		X							
50100C	.	H	.		O		X							
60100C	.	H	.		O		X							
73100	-5555	H	.		O		X				102.5	7.2	3.6	0
80100	-5555	H	.		O		X				105.3	0.8	0.5	0
80100	-5555	H	.		O		X				101.8	8.9	4.4	0
80200	-5555	H	.		O		X				100.3	13.3	6.4	0
80200	-5555	H	.		O		X				98.0	18.5	8.1	0
80200	-5555	H	.		O		X				101.9	14.9	9.1	0
80200	-5555	H	.		O		X				97.4	19.6	8.5	0
80300	-5555	H	.		O		X				105.5	0.3	0.2	0
80300	-5555	H	.		O		X				101.9	8.4	4.1	0
80400	-5555	H	.		O		X				105.5	0.5	0.3	0
80400	-5555	H	.		O		X				102.1	8.1	4.1	0
80500	-5555	H	.		O		X				103.5	4.8	2.5	0
80500	-5555	H	.		O		X				101.1	11.0	5.3	0
80700	-5555	H	.		O		X				100.2	14.3	7.0	0
80700	-5555	H	.		O		X				105.1	0.4	0.3	0
80700	-5555	H	.		O		X				100.0	12.8	5.8	0
80800	-5555	H	.		O		X				104.8	0.7	0.4	0
80900	-5555	H	.		O		X				107.0	0.6	0.3	0
80900	-5555	H	.		O		X				100.3	14.8	7.5	0
80900	-5555	H	.		O		X				103.7	4.6	2.4	0
81000	-5555	H	.		O		X				99.5	15.1	7.0	0
81000	-5555	H	.		O		X				102.0	8.5	4.3	0
81100	-5555	H	.		O		X				99.8	14.1	6.5	0
81100	-5555	H	.		O		X				102.3	8.0	4.1	0
81200	-5555	H	.		O		X				102.7	6.2	3.1	0
81200	-5555	H	.		O		X				105.9	0.2	0.1	0
81300	-5555	H	.		O		X				102.6	6.4	3.2	0
81300	-5555	H	.		O		X				100.4	14.0	7.0	0
81300	-5555	H	.		O		X				100.0	14.3	7.0	0
81300	-5555	H	.		O		X				100.7	14.3	7.4	0
81400	-5555	H	.		O		X				103.5	4.0	2.0	0
81400	-5555	H	.		O		X				101.6	8.9	4.3	0
81500	-5555	H	.		O		X				105.3	0.6	0.3	0
81500	-5555	H	.		O		X				107.6	0.2	0.1	0
RAW DATA		EGT		0...10...20...OATL..40...50...60						N2				ALT
DATE	CSI			0...10...20...MAR..40...50...60						0....5...MAR..15...20				MAINT CODES CTR
81200	-5555	H	.		O		X							0
81200	-5555	H	.		O		X							0
81300	-5555	H	.		O		X							0
81300	-5555	H	.		O		X							0
81300	-5555	H	.		O		X							0
81300	-5555	H	.		O		X							0
81400	-5555	H	.		O		X							0
81400	-5555	H	.		O		X							0
81500	-5555	H	.		O		X							0
81500	-5555	H	.		O		X							0

OAT'L CONCEPT & COMPUTATION



EGTM / OAT'L BASIC FORMULA

$$EGTMargin(\text{°C}) = 123315 - \left(\frac{EGT + 27315}{(\theta_0)^x} + K \left(NlKpm - \frac{(N)}{(\theta_0)^{0.47}} \right) \right) (\theta_{peak})$$

$$\text{Where: } \theta_0 = \frac{(TAT + 273,15)}{288,15}$$

$K = EGT$ (°C) per % NI

θ_{peak} = Hot Day Theta

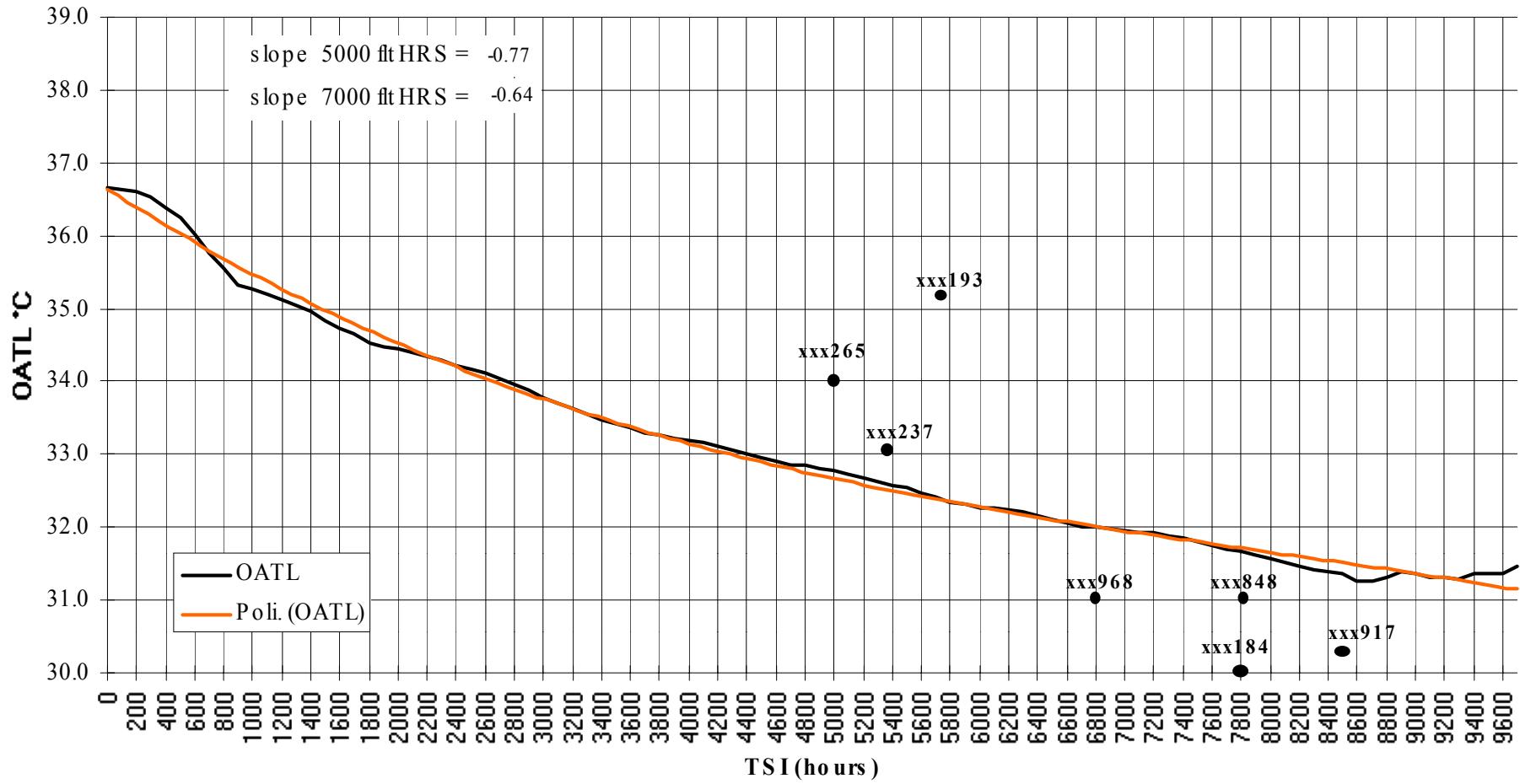
X = EGT exponential factor by engine model

$$OATL(\text{°C}) = \left(\frac{EGT_{RL}}{EGT_{RL} - EGTMargin} \right)^{1/0.95} \cdot (CP_{HD} + 273.15) - 273.15$$

Engine Rating	K	NlKpm%		(θ_{peak})	CP_{HD}	$Xof(\theta)$
		ON	OFF			
CF6-80C2B6F	81	106.55	107.55	1.058850	30.0	0.921

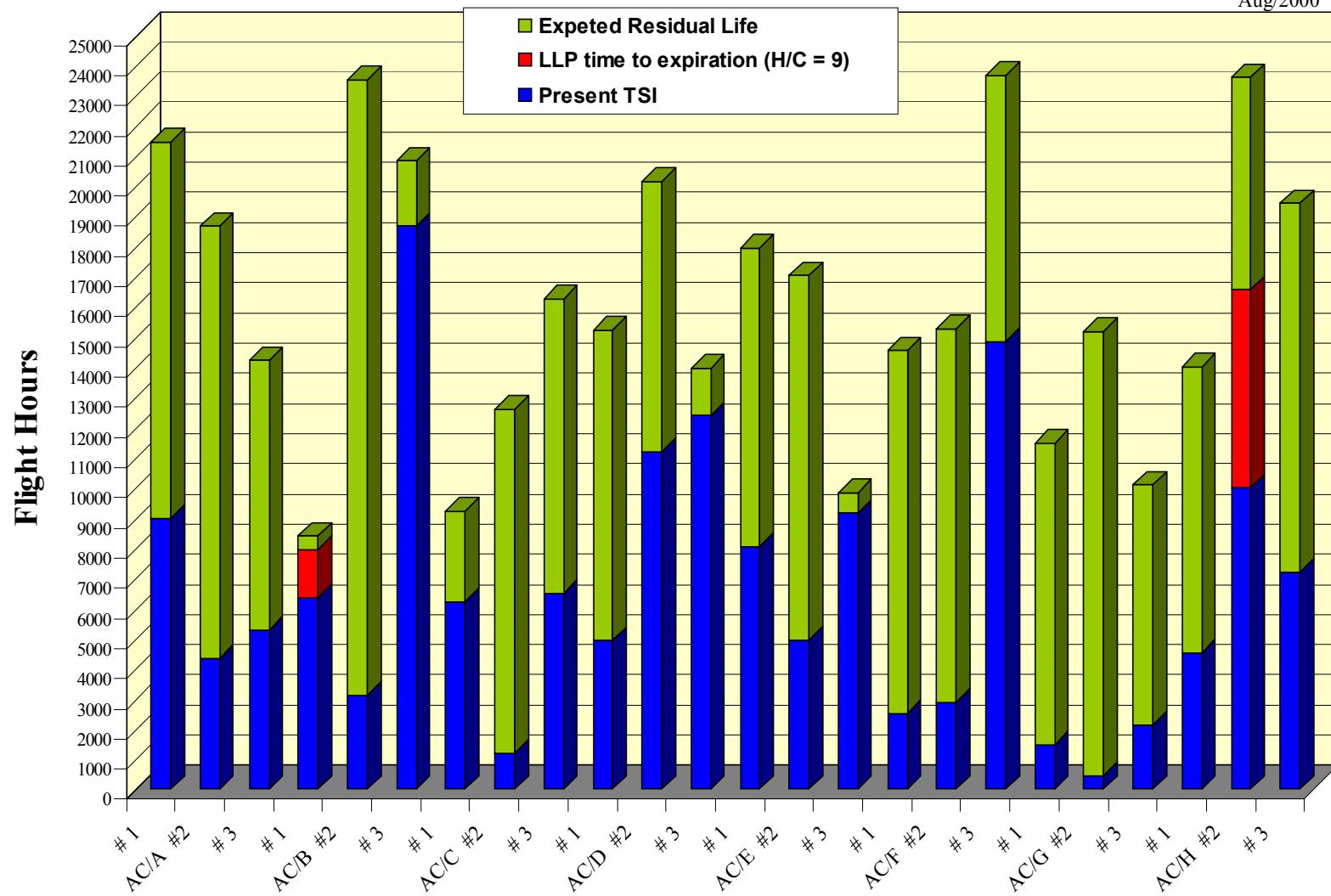
B747 ENG FLEET - OATL TREND
(qty75 sample engs based)

total slope -0.52

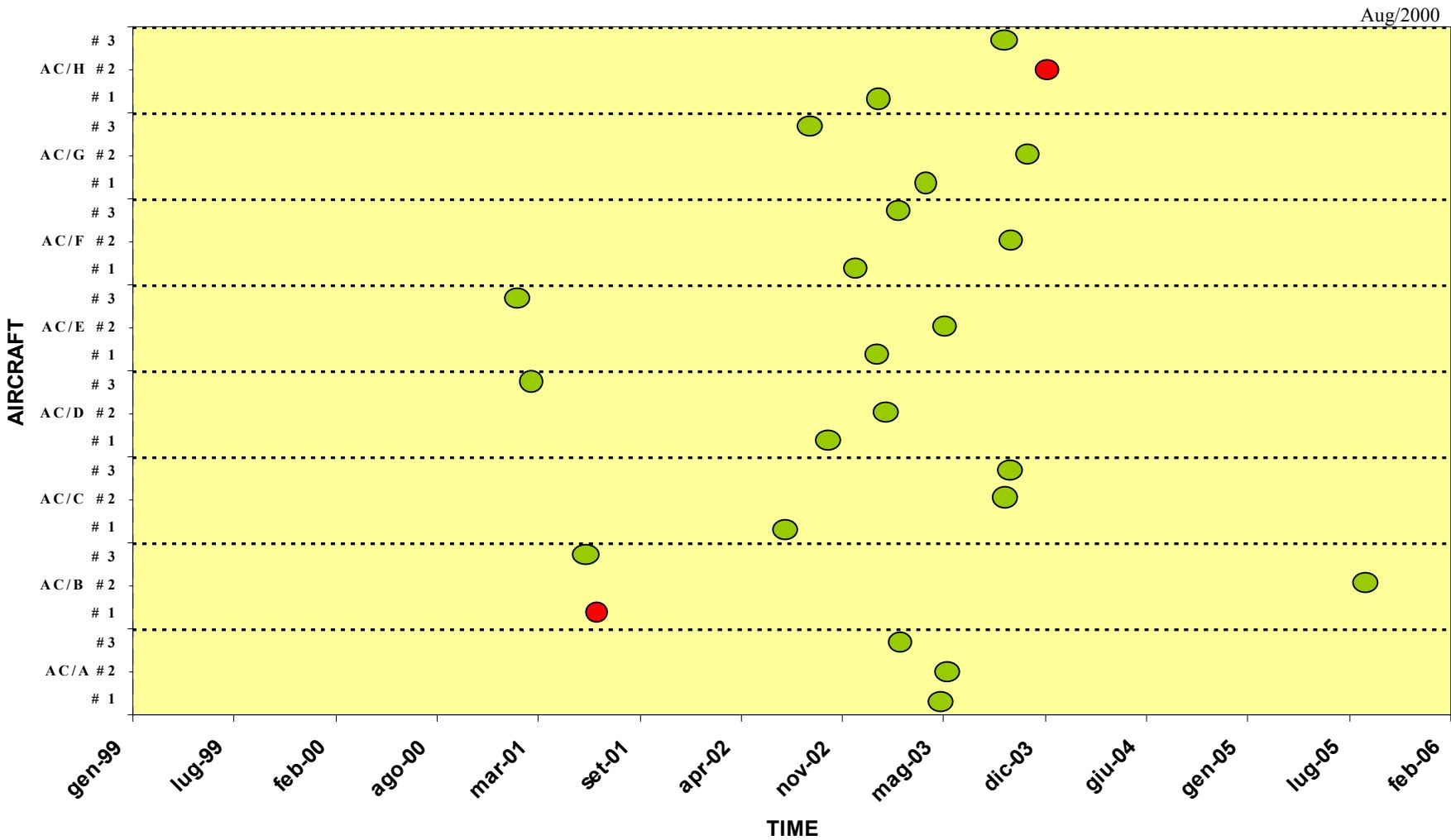


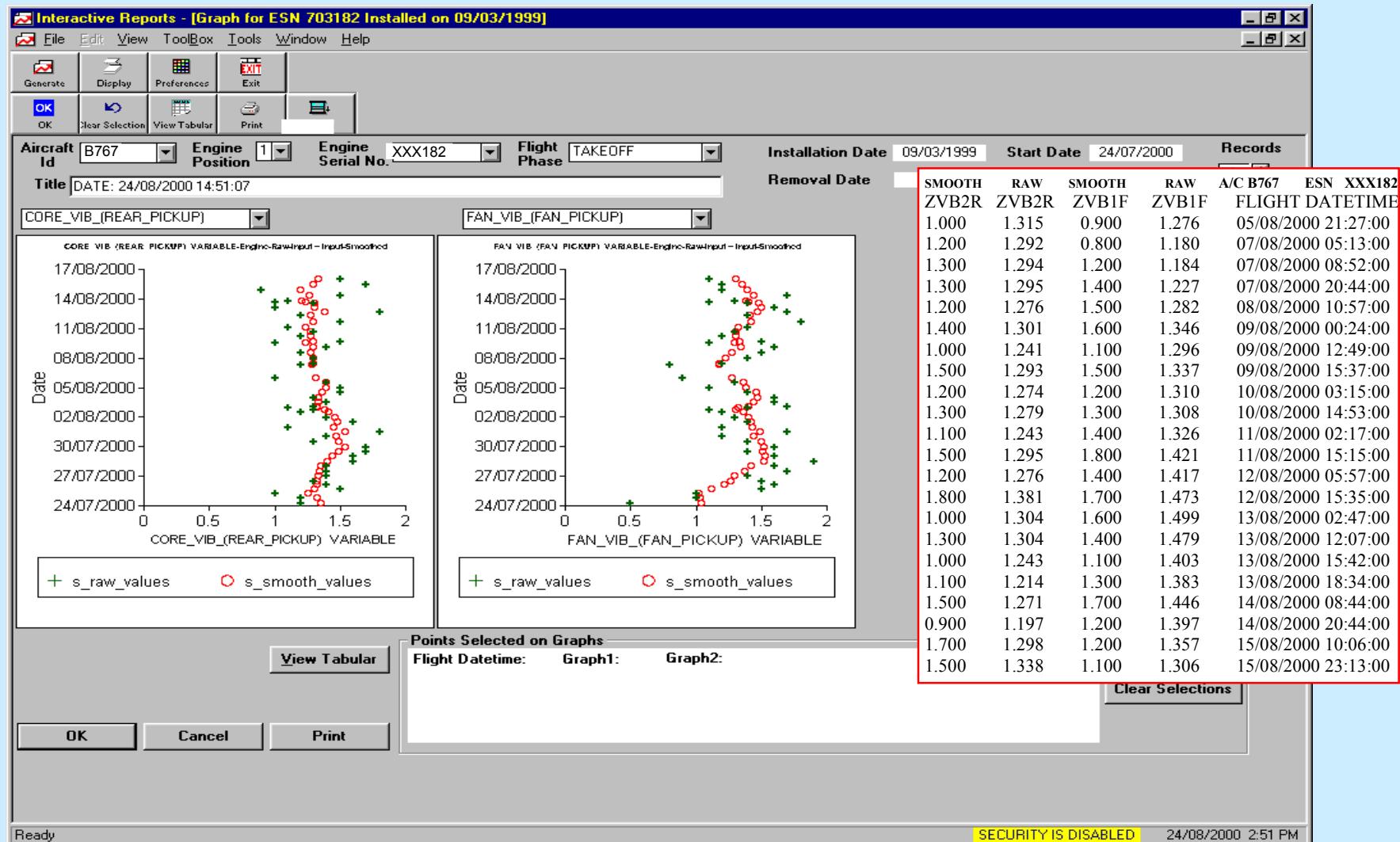
Alitalia - MD11 Fleet
 Individual Engine / Time to Shop Visit (Forecast)

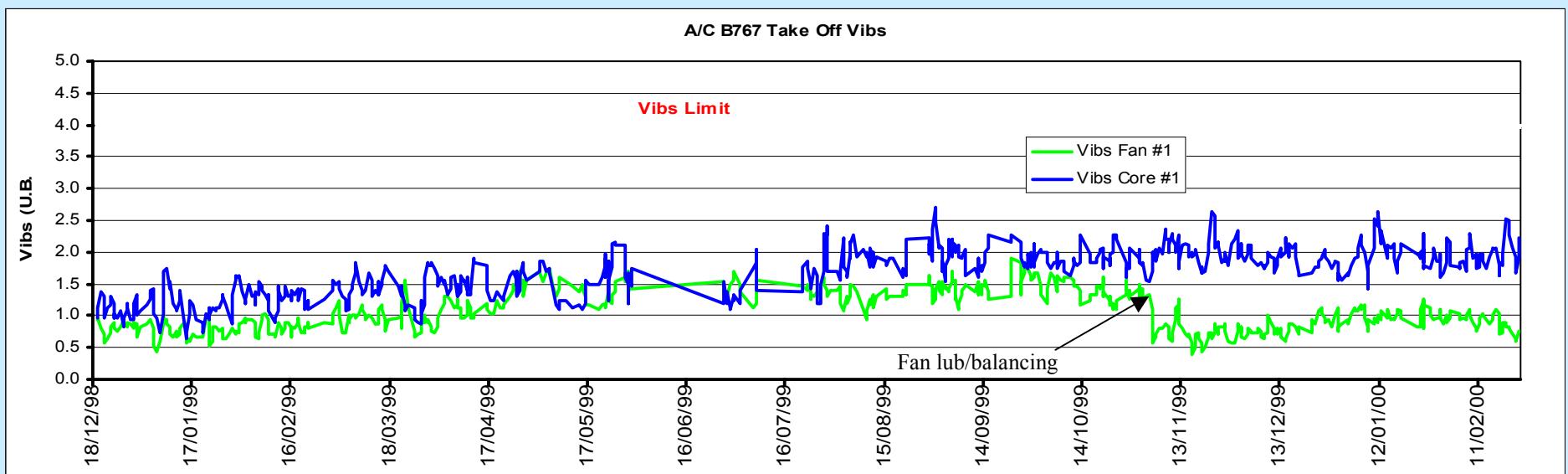
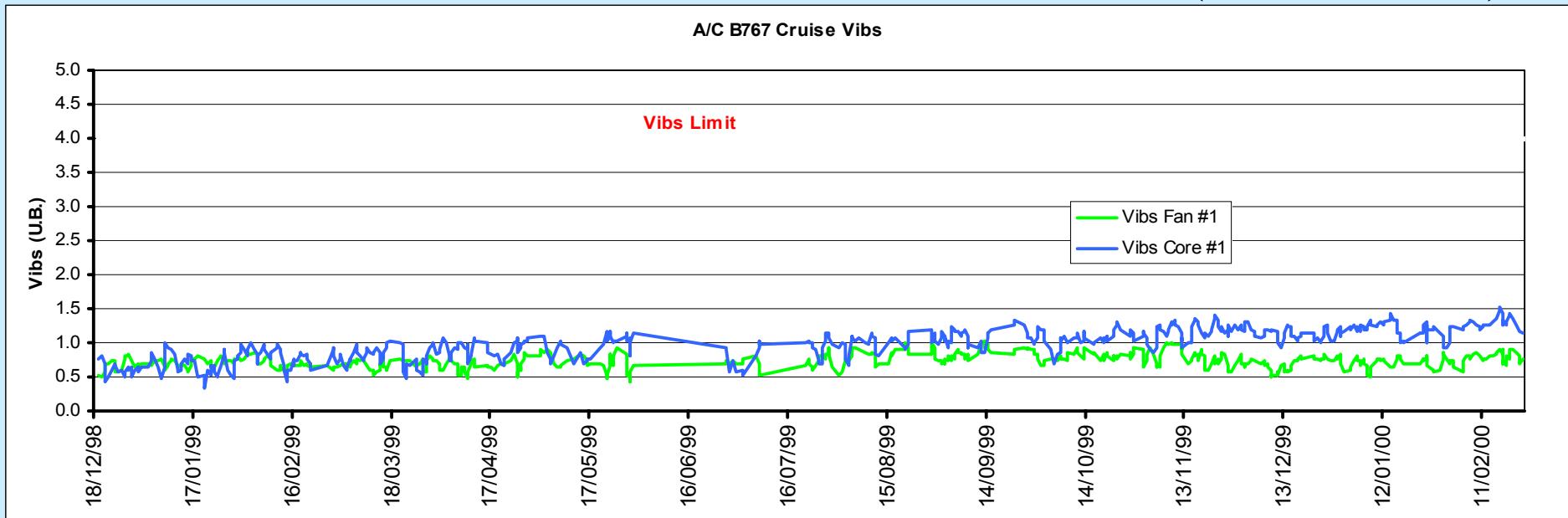
Aug/2000

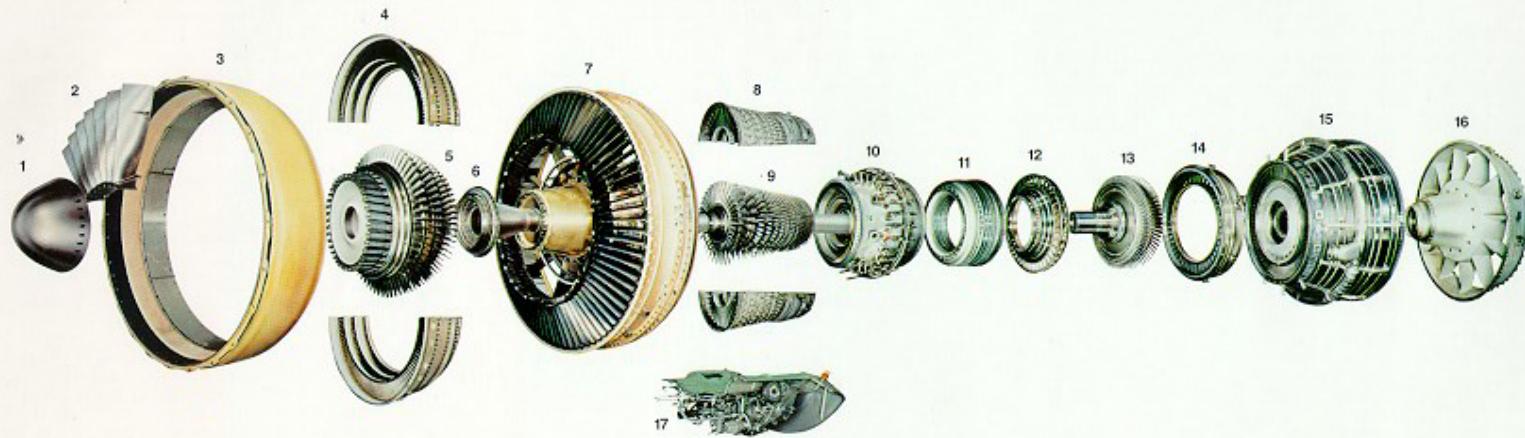


Alitalia - MD11 Fleet Engine Removals scenario (Expected)







*GE CF6-80C2 High Bypass Turbofan Engine*

Legend

- | | |
|---|--|
| 1. Spinner cone | 10. Compressor rear frame |
| 2. Fan blades (7 of 38) | 11. Combustor |
| 3. Fan forward case | 12. High pressure turbine (HPT) stage 1 nozzle |
| 4. Low pressure compressor (LPC) booster stator | 13. HPT rotor |
| 5. LPC booster rotor | 14. HPT stage 2 nozzle |
| 6. Fan shaft | 15. Low pressure turbine (LPT) |
| 7. Fan frame, aft case and outlet guide vanes | 16. Turbine rear frame |
| 8. High pressure compressor (HPC) stator | 17. Accessory gearbox and fire shield |
| 9. HPC rotor | |



ENGINES FLEET MAIN BUILD GROUPS SHOP MAINTENANCE THRESHOLDS

